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# FAILURE MODES EFFECTS ANALYSIS (FMEA) -- NON-CIL HARDWARE NUMBER:05-6PK-20310A -X

SUBSYSTEM NAME: EPD&C-COMMUNICATION & TRACKING:CLOSED CIRCUIT TV REVISION: 0

05/31/00

**PART DATA** 

PART NAME **VENDOR NAME** 

PART NUMBER **VENDOR NUMBER** 

LRU

:PANEL A7A1

V070-730356

SRU

:RESISTOR, CURRENT LIMITING

RWR80S1211FR

# **EXTENDED DESCRIPTION OF PART UNDER ANALYSIS:**

RESISTOR, CURRENT LIMITING, ½ WATT, 1.2 K-OHMS.

REFERENCE DESIGNATORS: 36V73A7A1A8R1

QUANTITY OF LIKE ITEMS: 1

**FUNCTION:** 

PROVIDE CURRENT LIMITING PROTECTION FROM SHORT IN SWITCH S57 OR THE CONNECTED WIRING.

REFERENCE DOCUMENTS: ECN 105-25016B DATED 2/25/99

# FAILURE MODES EFFECTS ANALYSIS FMEA -- NON-CIL FAILURE MODE

NUMBER: 05-6PK-20310A-02

REVISION#: 0

05/31/00

SUBSYSTEM NAME: EPD&C-COMMUNICATION & TRACKING:CLOSED CIRCUIT TV

LRU: PANEL A7A1

CRITICALITY OF THIS

ITEM NAME: RESISTOR, CURRENT LIMITING

FAILURE MODE: 1R3

**FAILURE MODE:** 

SHORT END-TO-END

MISSION PHASE:

PL PRE-LAUNCH

LO LIFT-OFF

OO ON-ORBIT

DO DE-ORBIT

LS LANDING/SAFING

VEHICLE/PAYLOAD/KIT EFFECTIVITY:

102 COLUMBIA

103 DISCOVERY

104 ATLANTIS

105 ENDEAVOUR

#### CAUSE:

STRUCTURAL FAILURE(MECHANICAL STRESS, VIBRATION), ELECTRICAL STRESS, THERMAL STRESS, PROCESSING ANOMALY

CRITICALITY 1/1 DURING INTACT ABORT ONLY? NO

REDUNDANCY SCREEN

A) PASS

B) PASS

C) PASS

### **PASS/FAIL RATIONALE:**

A)

SHORT END-TO-END FAILURE OF RESISTOR DETECTABLE DURING GROUND TURNAROUND USING BREAKOUT BOX.

SHORT END-TO-END DETECTABLE IN ORBIT AS THIS FAILURE WOULD RESULT IN LOSS OF CONTROL BUS BC1.

C)

### - FAILURE EFFECTS -

#### (A) SUBSYSTEM:

LOSS OF CURRENT LIMITING PROTECTION TO THE CONTROL BUS BC1.

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## (B) INTERFACING SUBSYSTEM(S):

POSSIBLE EFFECT ON ANY INTERFACING SYSTEMS CONTROLLED BY BUS BC1.

#### (C) MISSION:

POSSIBLE LOSS OF MISSION AFTER THREE FAILURES SEE (D) FOR SCENARIO.

## (D) CREW, VEHICLE, AND ELEMENT(S):

POSSIBLE LOSS OF CREW/VEHICLE AFTER THREE FAILURES:

- (1) RESISTOR R1 SHORTS END-TO-END
- (2) TOGGLE SWITCH S57 SHORTS TO CASE RESULTING IN POSSIBLE DAMAGE TO CONTROL BUS BC1. ALL CRITICAL FUNCTIONS ON BUS BC1 HAVE BACKUP.
- (3) LOSS OF NEXT CONTROL BUS MAY CAUSE LOSS OF CREW/VEHICLE.

### (E) FUNCTIONAL CRITICALITY EFFECTS:

SHORT END TO END OF RESISTOR R1 MAY CAUSE LOSS OF CREW/VEHICLE.

## - APPROVALS -

SSS&R ENGINEERING DESIGN ENGINEERING

: K.E.RYAN/C.S.PUTCHA

: G.J.SCHWARTZ